

INSTITUTE OF AERONAUTICAL ENGINEERING

(Autonomous) Dundigal, Hyderabad - 500 043

MECHANICAL ENGINEERING

ASSIGNMENT QUESTIONS

Course Name	:	FINITE ELEMENT METHODS
Course Code	:	A60330
Class	:	III YEAR II SEMESTER
Branch	:	MECHANICAL ENGINEERING
Year	:	2017 – 2018
Course Faculty	:	Prof V. V. S. H. Prasad, Professor
-		Mr. C. Labesh Kumar, Assistant Professor.

OBJECTIVES

The aim of this course is to introduce basic principles of Numerical methods and it is further extended to cover the application of finite element method by the inclusion of 1D bar elements truss elements beam elements 2D elements like CST, ring elements. Nowadays the principles of finite element method find wide applications in many engineering fields like mechanical engineering, aerospace, civil engineering, nuclear engineering, bio mechanics etc. The course deals with the Raleigh-ritz (PMPE) and galerkin approaches. The objective of the course is determined structural deformations strains element stress and heat transfer problems.

S. No	Question	Blooms Taxonomy Level	Course Outcome		
ASSIGNMENT I					
1	Using potential energy approach, describe FE formulation for plane truss Element.	Understand	1		
2	An axial load P=300X10 ³ N is applied at 20 ⁰ C to the rod as shown in Figure below. The temperature is the raised to 60^{0} C. a) Assemble the K and F matrices. b) Determine the nodal displacements and stresses. $\begin{array}{cccccccccccccccccccccccccccccccccccc$	Understand	1		
3	The tripod shown in figure below carries a vertically downward load of 10kN at joint 4. If Young's modulus of the material of tripod stand is 200kN/mm ² , determine the forces developed in the legs of the tripod.	Remember	2		

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4	For the beam shown in Figure below, determine the following:	Remember	2
	a) Slopes at nodes 2 and 3.		
	b) Vertical deflection at the mid-point of the distributed load. Consider		
	have $E=200$ GPa, $I=5X10^6$ mm ⁴ .		
	24 kN/m		
	$(1) 2 \xrightarrow{\text{from}} (2) 3 \xrightarrow{\text{from}}$		
	← 1 m → ← 1 m →		
	Figure beam with distributed load		
5	A beam fixed at one end and supported by a roller at the end, has a 20KN concentrated load applied at the centre of the span, as shown in		
	20 Kin concentrated to a applied at the centre of the span, as shown in fig. calculate the deflection under the load and construct shear force and		
	bending moment diagram for the beam.		
	Take $E = 20 \times 10^6 \text{ N/c}$, ² I=2500 cm ⁴ .	Understand	2
	(1) (2)		
	A SSICNMENT-II		
1	For the configuration shown in figure determine the deflection at the		
1	point load application		
	using a one element model. $T = 10 \text{ mm}$, $E = 70 \text{ G Pa}$, $v = .3$		
	100 N	Pamambar	1
	30 mm 50 N	Remember	1
	20 mm		
2	a)A four node quadrilateral element is shown in fig 1.3 the co-ordinates		
2	of each node are given in cm. The element displacement vector is given		
	as [q]=[0 0 0.2 0 0.15 0.10 0 0.05] cm. find (i) the x, y co-ordinates of a		
	point P whose location at $\xi = 0.5$, $\eta = 0.5$ (ii) the displacement of point	Understand	3
	P(u,v) (iii)the jacobian at P b)Evaluate the Integral $I = \begin{bmatrix} 1 & (2\xi^2 \pm \xi^3) \\ 1 & (2\xi^2 \pm \xi^3) \end{bmatrix} d\xi$ using Gaussian quadrature		
	$r_{j-1} = (3\zeta + \zeta) \alpha \zeta$ using Gaussian quadrature method		
3	The plane wall shown in fig. The thermal conductivity $K = 25 W/m^0 c$ and		
	there is a uniform		
	generation of heat in the wall of $Q = 400$ W/m ³ . Determine the		
	nodes (include two sides of the walls) in equal distances through the wall		
	thickness.	Remember	2
			-
	200° c •		
	×		
4	a. Evaluate natural frequencies for the stepped har shown fig in axial		
-	vibration take $E=200$ GPa and DENSITY= 7850 Kg/m ³ .		
	b. Draw mode shapes and determine Eigen vector.	Remember	2
	Take A_1 =400mm ² and A_2 =200mm ² using characteristic polynomial		
	method.		

S. No	Question	Blooms Taxonomy Level	Course Outcome
	A1 A2 200 mm - 100 mm -		
5	Find the approximate first two natural frequencies of a simply supported beam using on a element. Flexural Rigidity =EI; Density = \mathcal{P} Cross- sectional area=A \mathbf{y} \mathbf{z} \mathbf{x}	Remember	4

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